

CODE's multi-GNSS orbit and clock solution - status 2016

L. Prange, E. Orliac, R. Dach, D. Arnold, G. Beutler,
S. Schaer, A. Jäggi

Astronomical Institute, University of Bern, Switzerland

EGU General Assembly, 17-22 April 2016,
Vienna, Austria



Astronomical Institute University of Bern



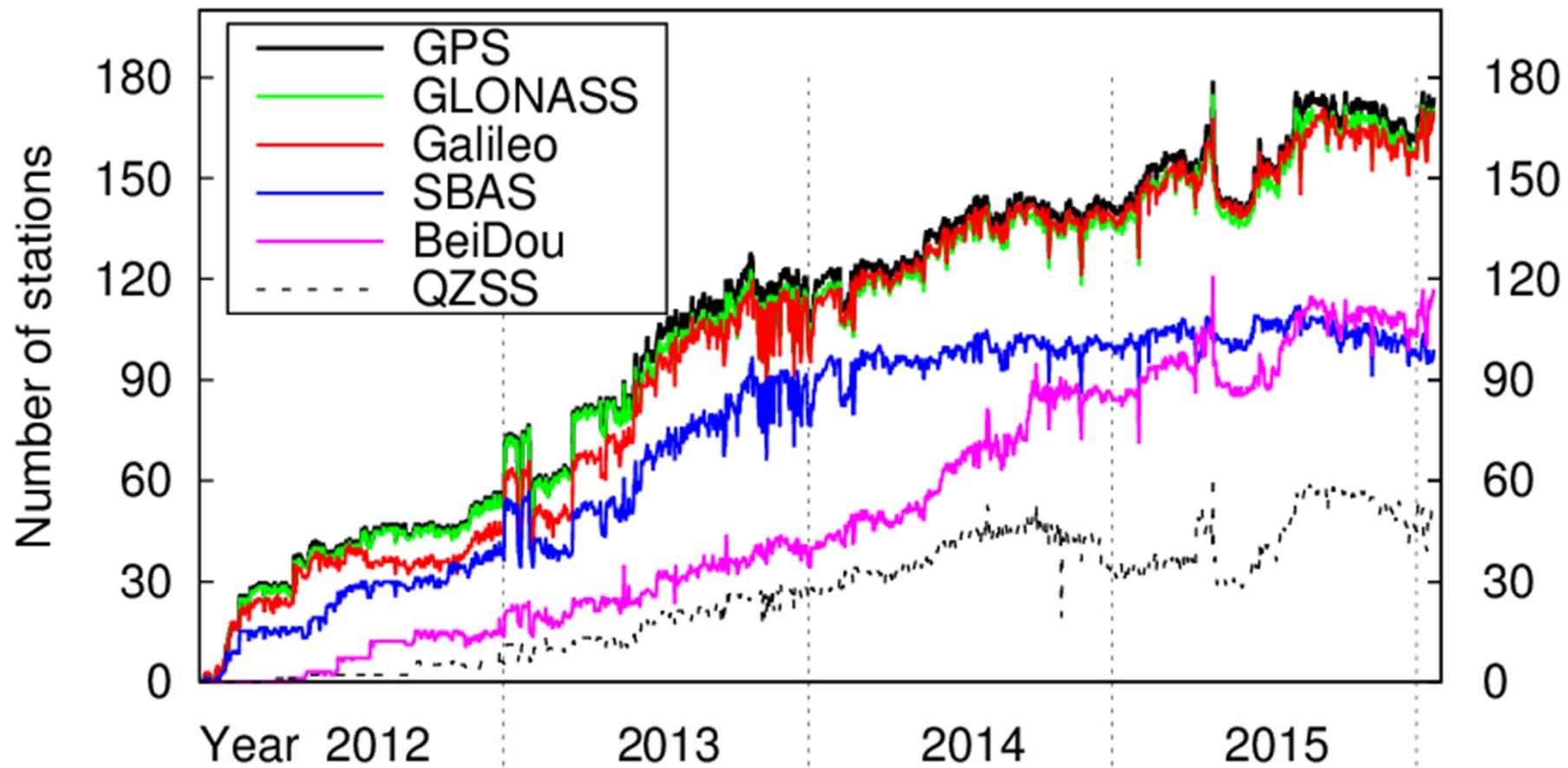
Contents

- Data base and network
 - CODE MGEX orbit solution
 - CODE MGEX clock solution
 - Model changes and implementations
 - Satellite orbit validation
 - Satellite clock validation
 - Summary and outlook
-



MGEX data monitoring

Number of stations providing daily RINEX3 files and included in CODE's raw data monitoring (data sources: IGS-MGEX and EPN)



CODE MGEX orbit solution

GNSS considered:	GPS + GLONASS + Galileo + BeiDou (MEO+IGSO) + QZSS (≈70 SV)
Processing mode:	Post-processing (≈2 weeks latency)
Timespan covered:	GPS-weeks 1689 - today
Number of stations:	130 (GPS), 110 (GLONASS), 85 (Galileo); 55 (BeiDou); 20 (QZSS)
Processing scheme:	Double-difference network processing (observable: phase double differences)
Signal frequencies:	L1+ L2 (GPS + GLO+ QZSS); E1 (L1) + E5a (L5) Galileo; B1 (L1) + B2 (L7) BeiDou
Orbit characteristic:	3-day long arcs; SRP: ECOM / ECOM2 (since 2015)
Reference frame:	IGS08 (until week 1708); IGb08 (since week 1709)
IERS conventions:	IERS2003 (until 1705); IERS2010 (since 1706)
Product list:	Daily orbits (SP3) and ERPs
Distribution:	ftp://cddis.gsfc.nasa.gov/gnss/products/mgex/ and ftp://ftp.unibe.ch/aiub/CODE_MGEX/
Designator:	comwwwwd.???.Z



CODE MGEX clock solution

GNSS considered:	GPS + GLONASS + Galileo + BeiDou + QZSS (≈ 70 SV)
Processing mode:	Post-processing (≈ 2 weeks latency)
Timespan covered:	GPS-weeks 1710 - today
Number of stations:	130 (GPS), 35 (GLO), 45 (Galileo); 50 (BeiDou); 20 (QZSS)
Processing scheme:	Zero-difference processing (observable: code+phase undifferenced)
Signal frequencies:	L1+ L2 (GPS + GLO+ QZSS); E1 (L1) + E5a (L5) Galileo; B1 (L1) + B2 (L7) BeiDou
A priori information:	Orbits, ERPs, coordinates, and troposphere from CODE MGEX orbit solution introduced as known
Reference frame:	IGb08
IERS conventions:	IERS2010
Product list:	Epoch-wise (300s) satellite and station clock corrections in daily clock RINEX files; daily inter-system biases for mixed stations in Bernese DCB and BIAS-SINEX format ftp://cddis.gsfc.nasa.gov/gnss/products/mgex/ and ftp://ftp.unibe.ch/aiub/CODE_MGEX/
Distribution:	

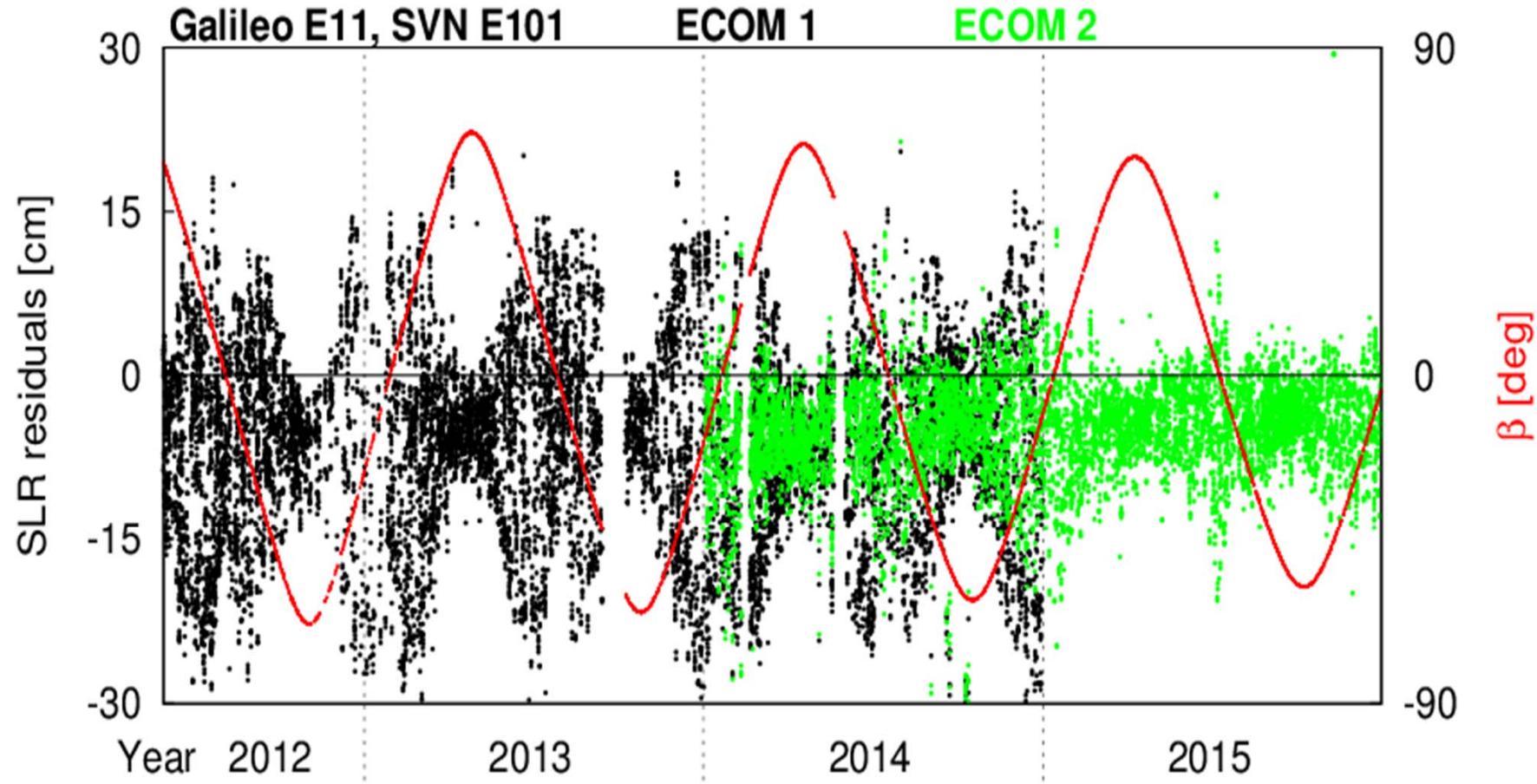


Changes w.r.t. to IGS solution

- Implementation of Galileo, QZSS, BeiDou (except GEOs)
- Use of RINEX3 files from MGEX network; selection of observation types
- Improved SRP model for yaw-steering attitude (ECOM2, Arnold et al., 2015)
- Normal attitude and related SRP models for QZSS and BeiDou
- ANTEX (PCO+PCV) for Galileo, QZSS, BeiDou
- Attitude laws for GPS, GLONASS, (Galileo?) eclipses
- Proper handling of observation biases
- Ambiguity resolution for Galileo, BeiDou, QZSS
- Albedo radiation modelling for Galileo, QZSS, BeiDou
- Antenna thrust for (GLONASS), Galileo, QZSS, BeiDou

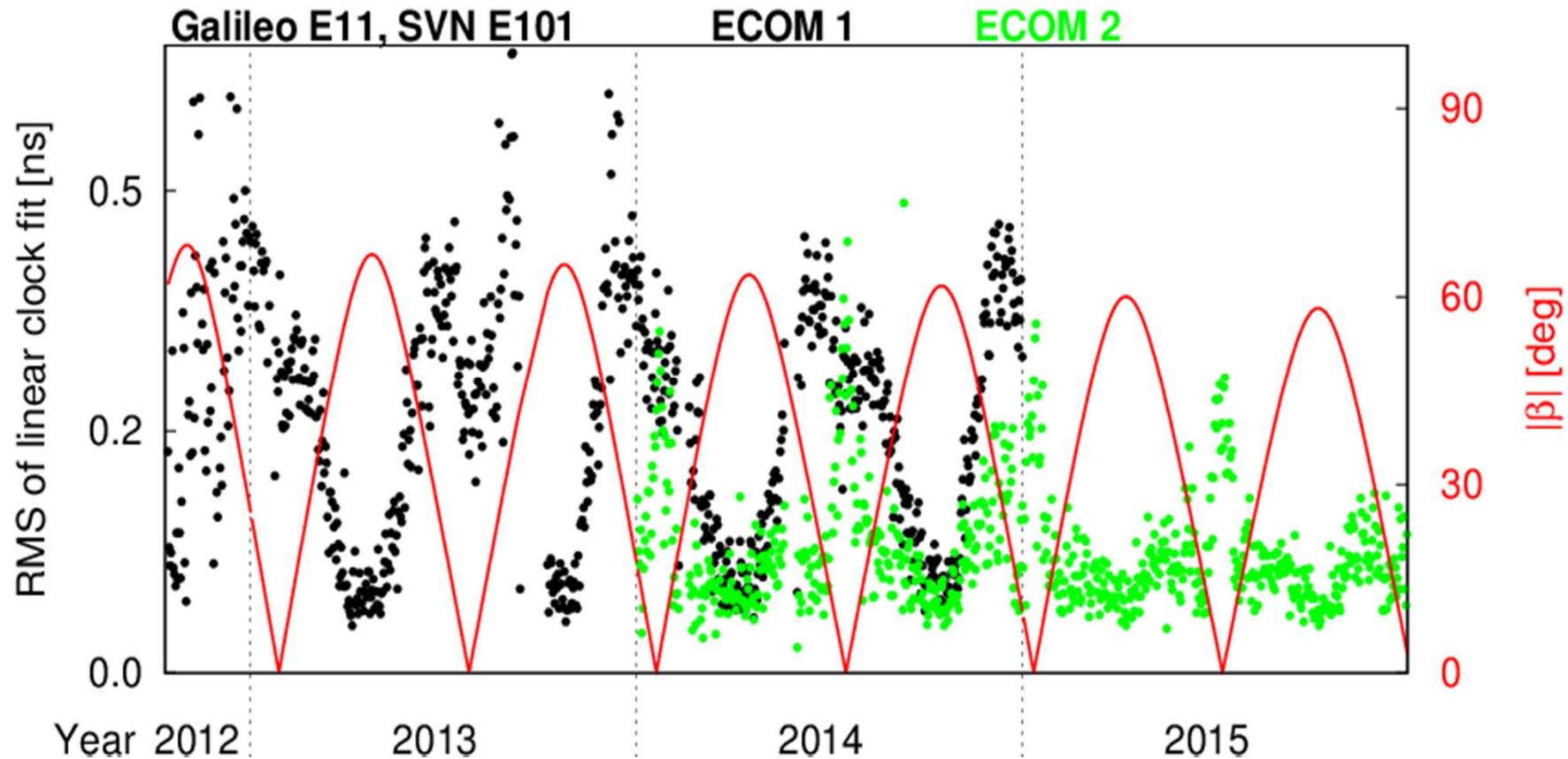


Galileo orbit validation



- Significant reduction of dependency on beta-angle, when changing to the ECOM2 (attitude related?)

Galileo clock validation

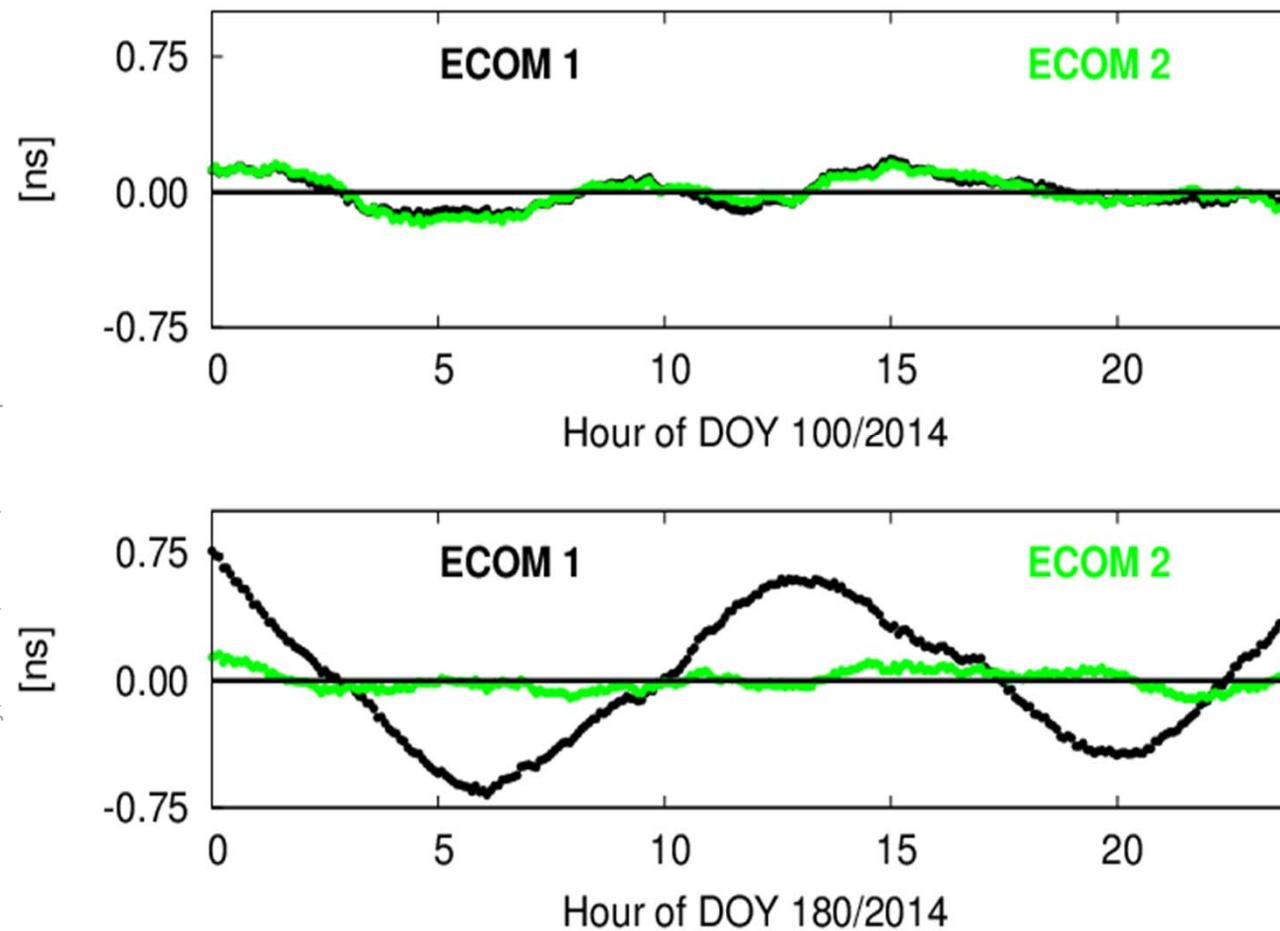


- Significant reduction of dependency on beta angle
- Pronounced signal remains during eclipse season or close-by (attitude related?)



Galileo clock validation

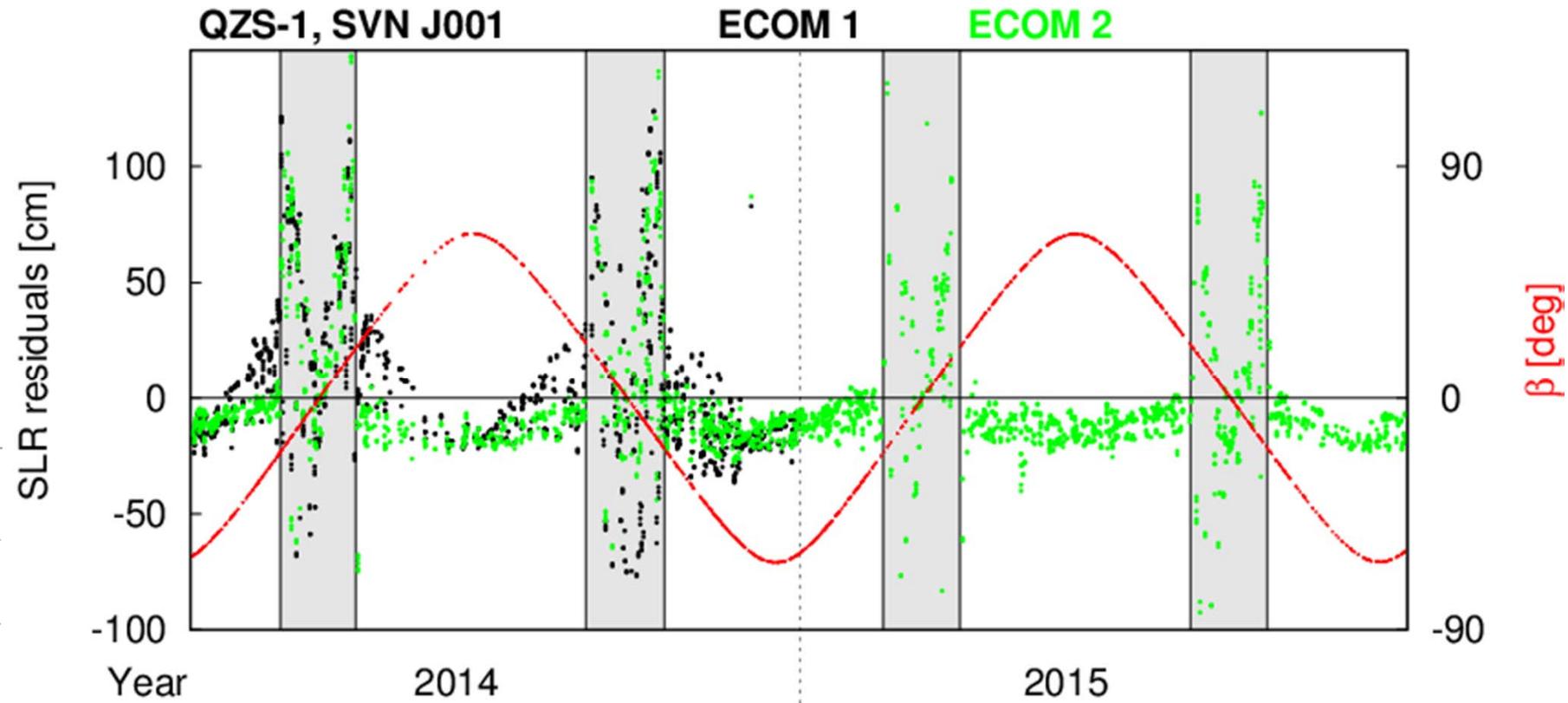
Clock corrections of Galileo E11, SVN E101



Large beta-angle:
=> Clock signal has
small amplitude
(about ± 0.15 ns)

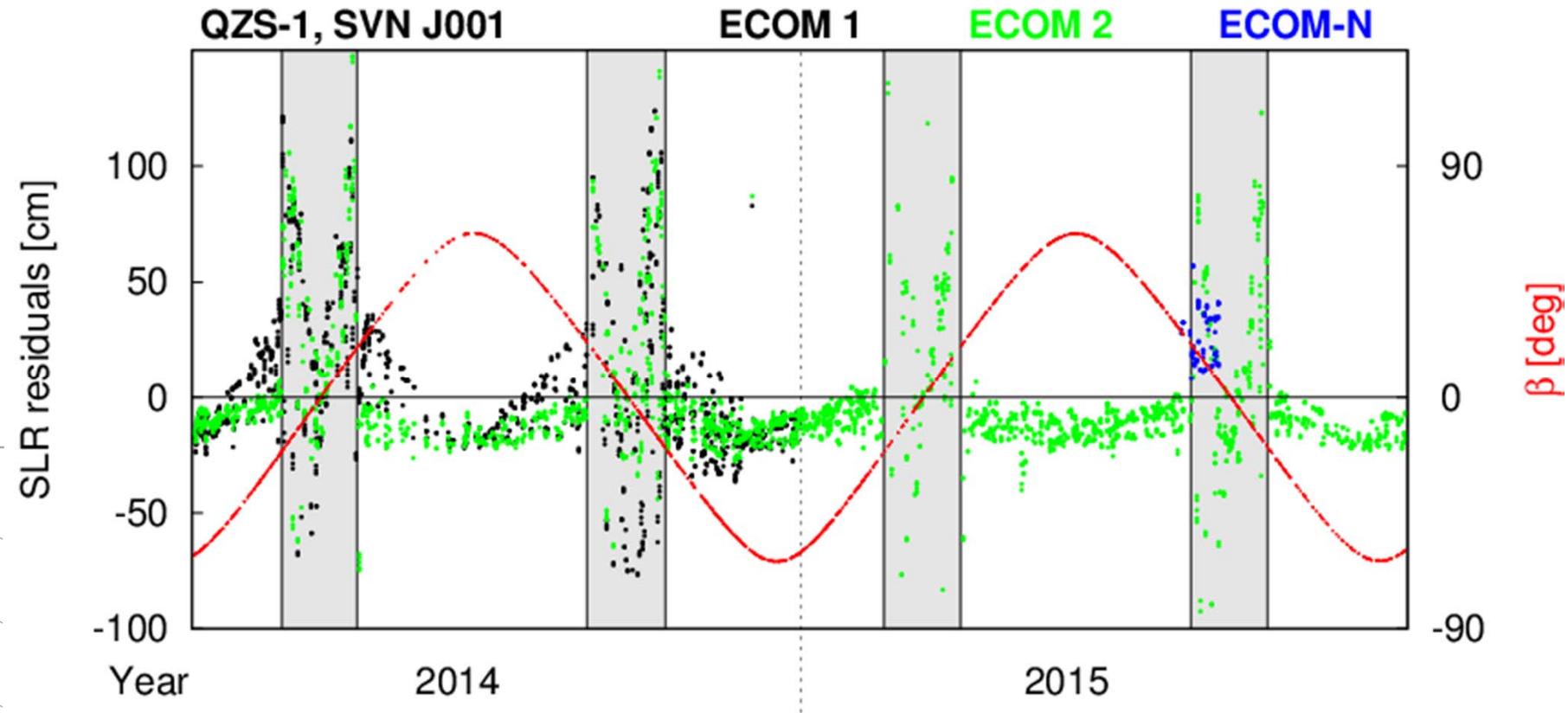
Small beta-angle:
=> Periodic signal
caused by mis-
modelled orbit
(ECOM1)
=> Significant reduction
of signal amplitude
from ± 0.75 ns to
 ± 0.15 ns when
switching to
ECOM2

QZSS orbit validation



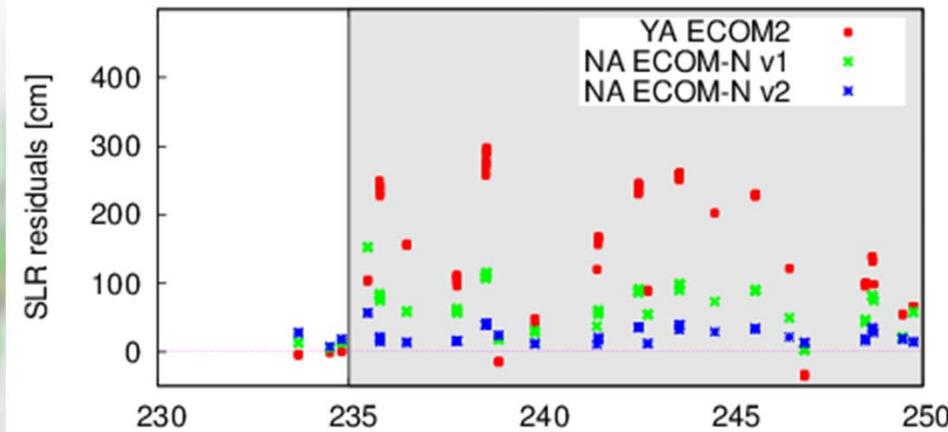
- ⇒ Yaw-steering: ECOM2 reduces dependency on beta angle
- ⇒ Significant SLR offset remains

QZSS orbit validation



- Normal attitude mode ($|\beta| < 20^\circ$; marked grey): large orbit errors
- Test of new ECOM versions suited for orbit normal attitude mode

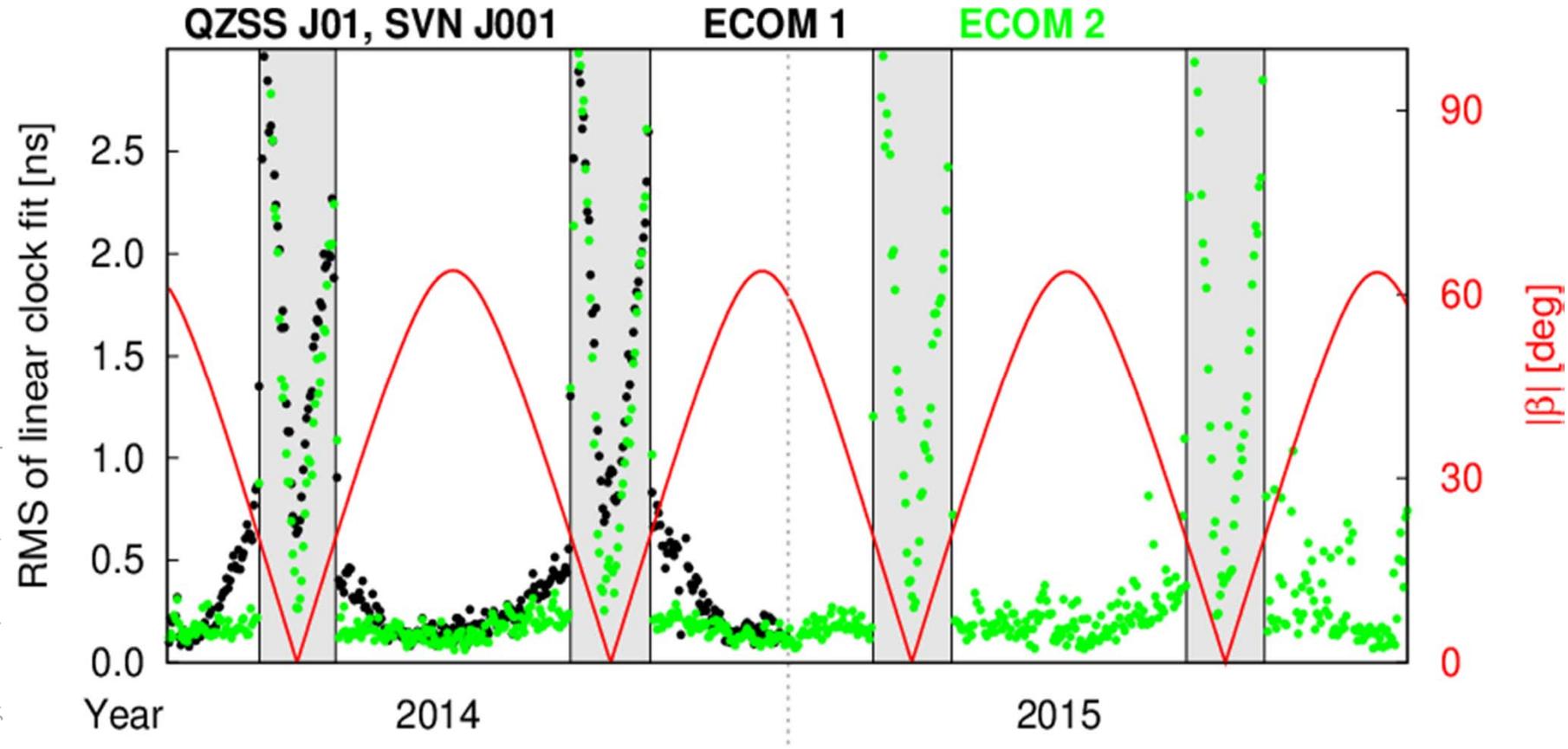
QZSS orbit validation



⇒ Test of new ECOM versions suited for orbit normal attitude mode

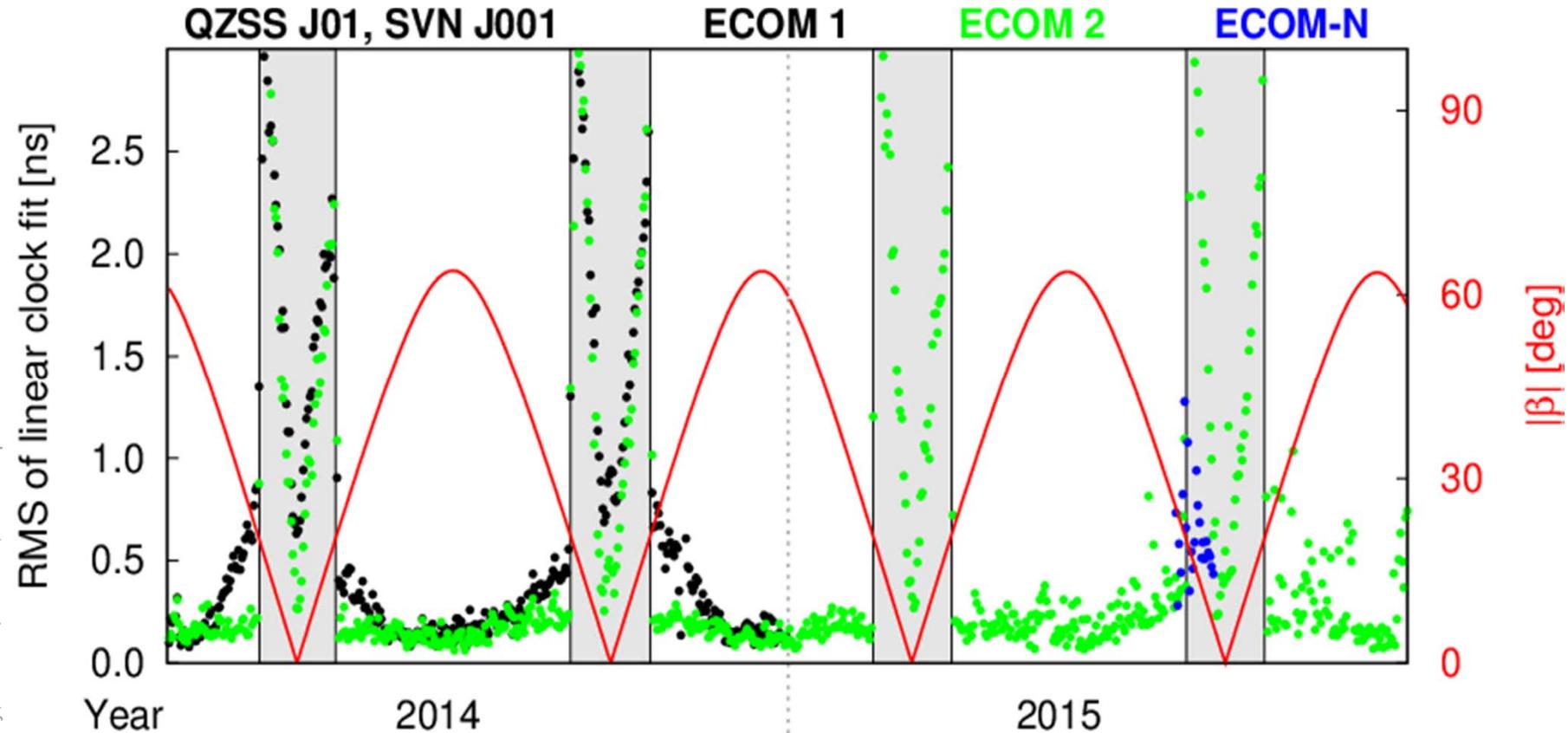


QZSS clock validation



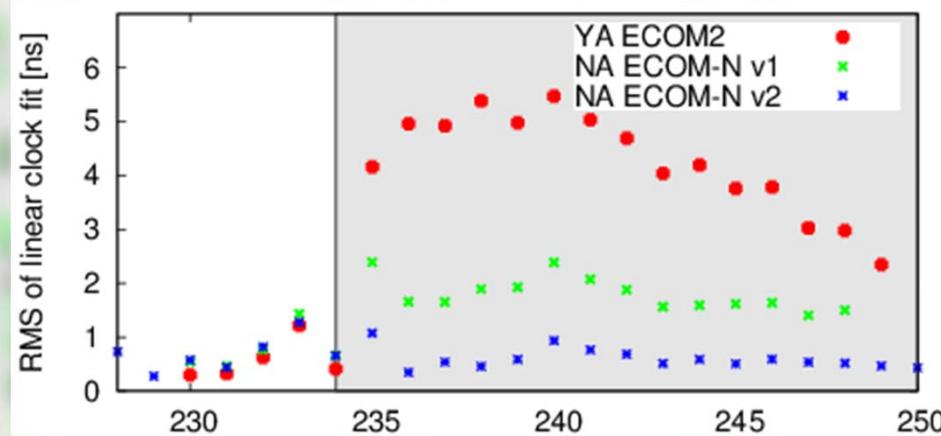
- ➡ Yaw-steering: significant reduction of dependency on beta angle thanks to ECOM2
- ➡ Orbit normal attitude mode (grey): large errors remain

QZSS clock validation



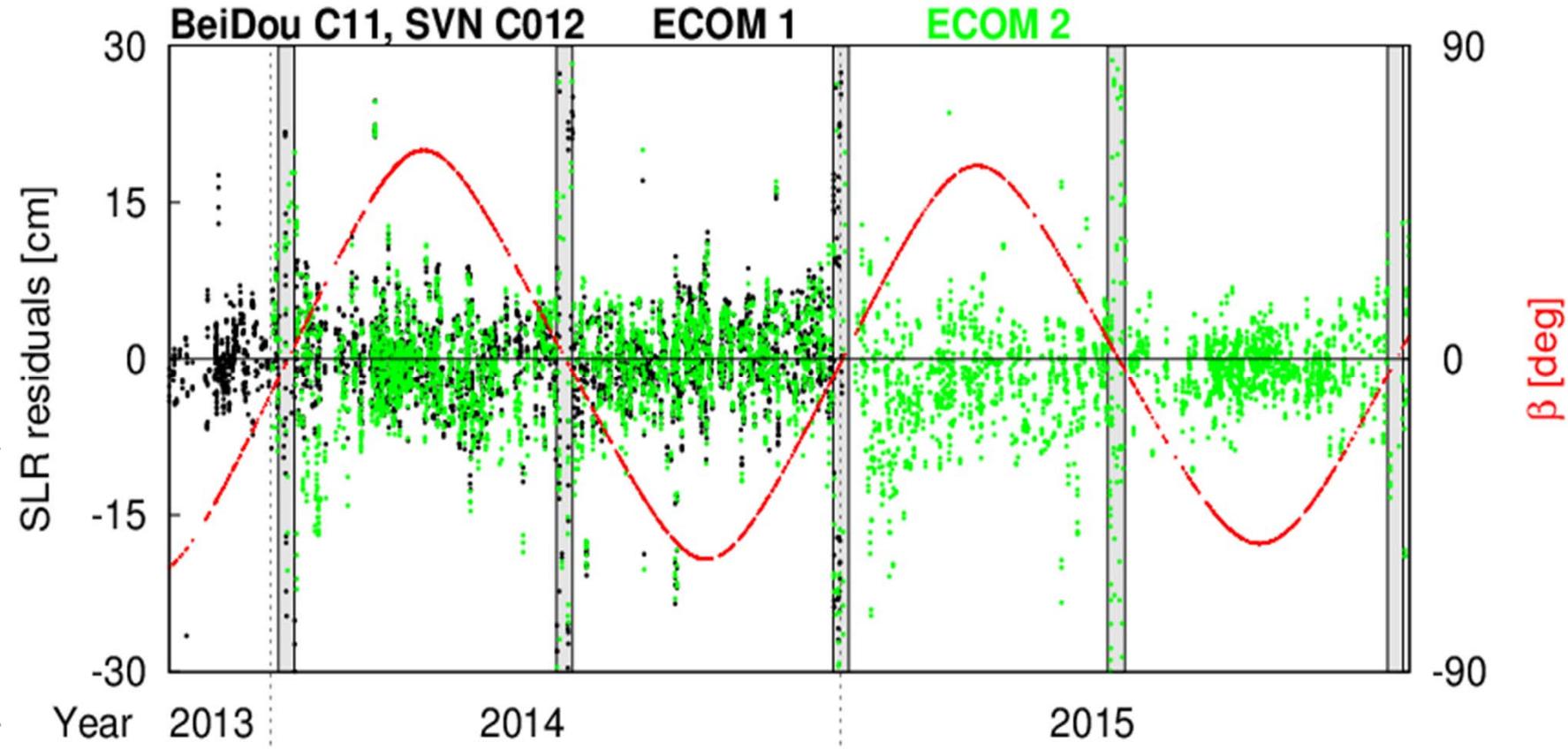
⇒ Experiments with ECOM versions better suited for orbit normal attitude mode

QZSS clock validation



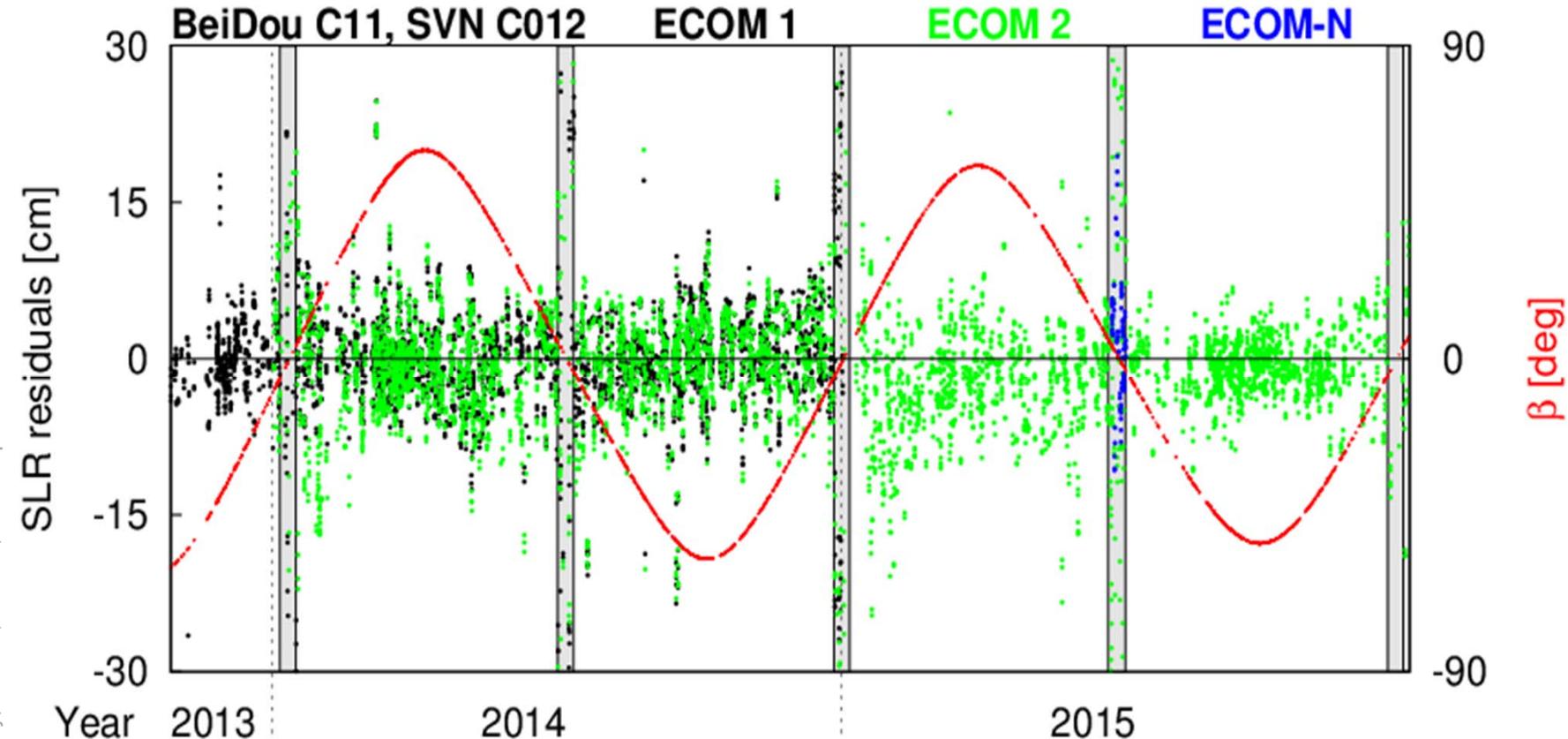
⇒ Experiments with ECOM versions better suited for orbit normal attitude mode

BeiDou orbit validation



- ➡ Yaw-steering: no significant impact of ECOM version
- ➡ Orbit normal attitude mode ($|\beta| < 4^\circ$; grey boxes; not correctly considered): large residuals

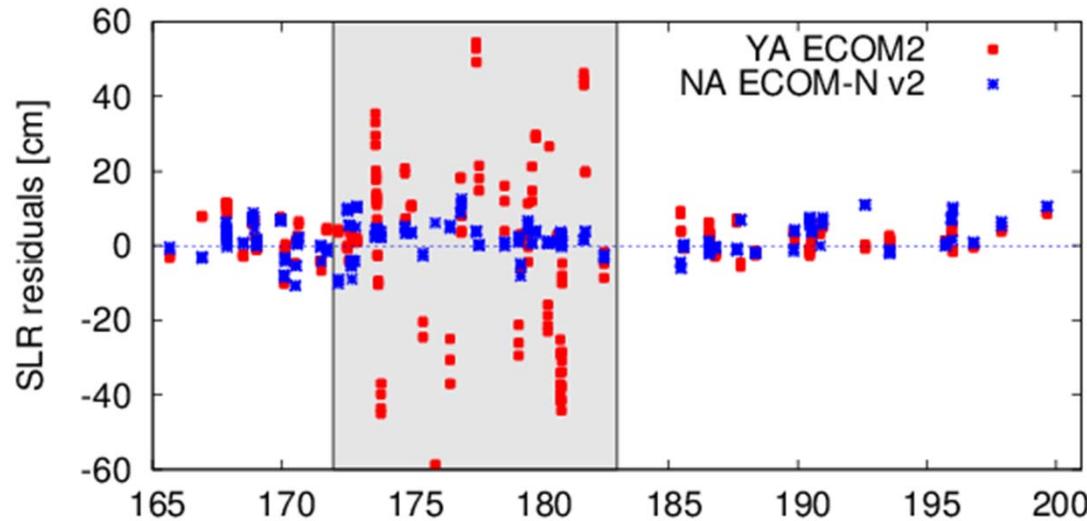
BeiDou orbit validation



→ Test of new ECOM versions better suited for orbit normal attitude mode



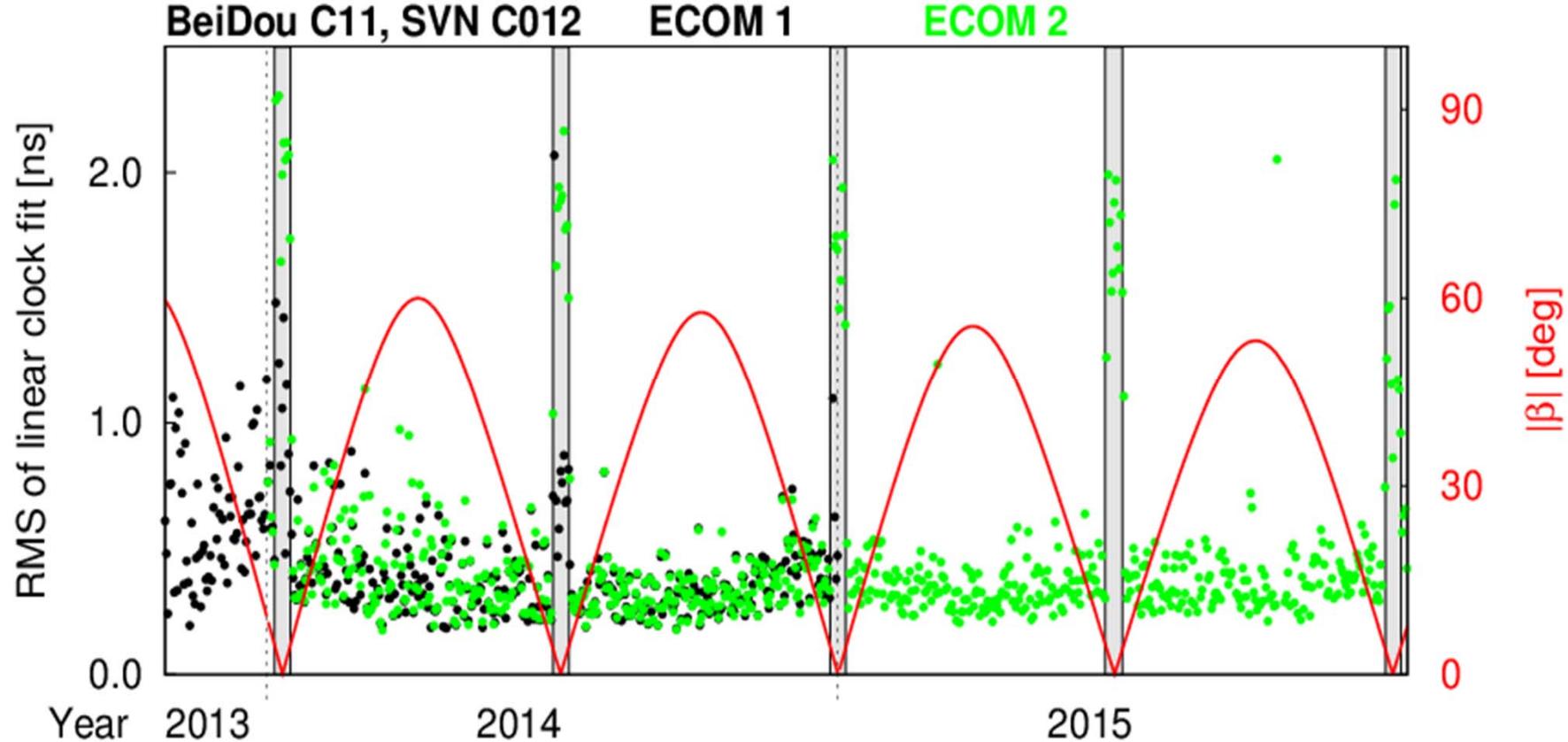
BeiDou orbit validation



⇒ Test of new ECOM versions better suited for orbit normal attitude mode

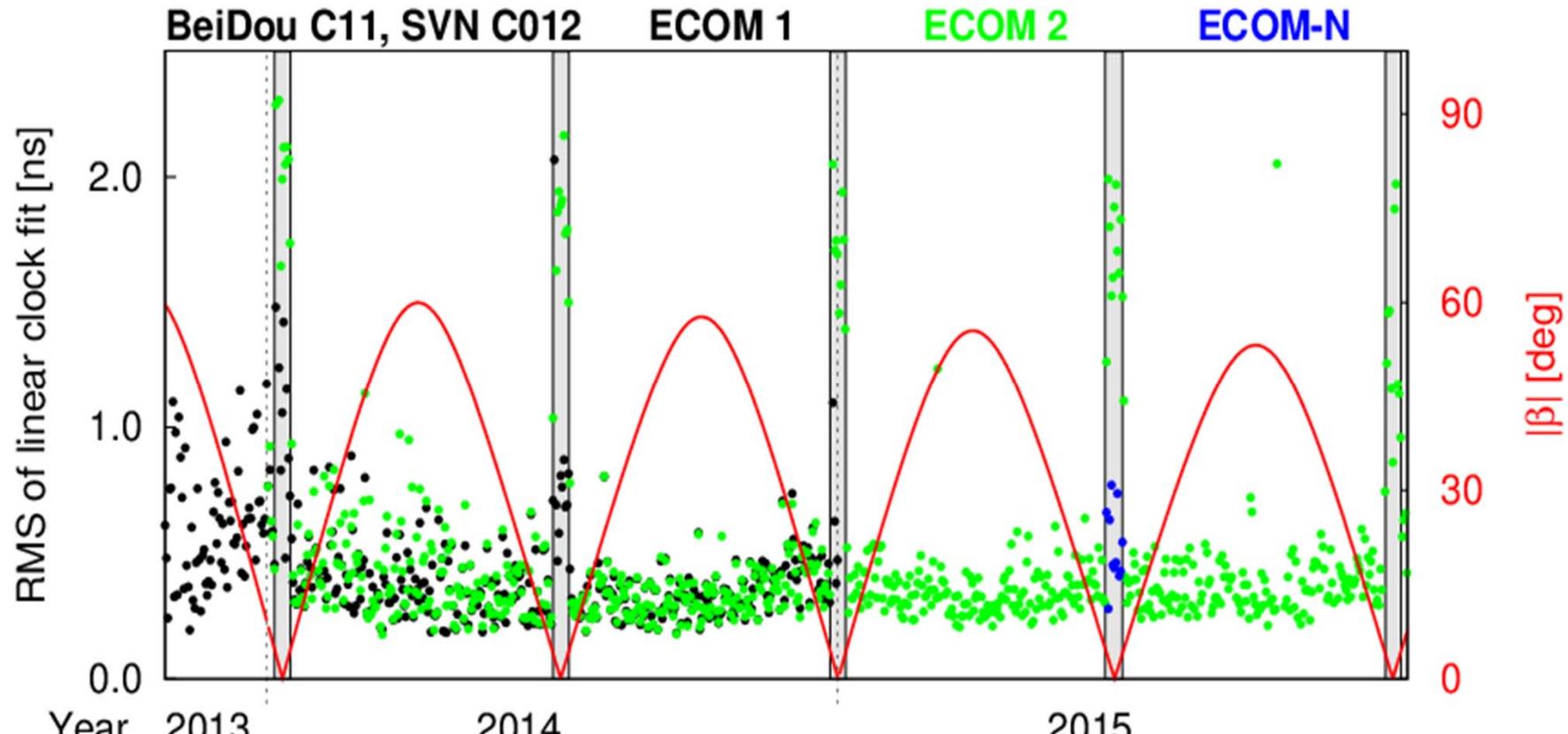


BeiDou clock validation



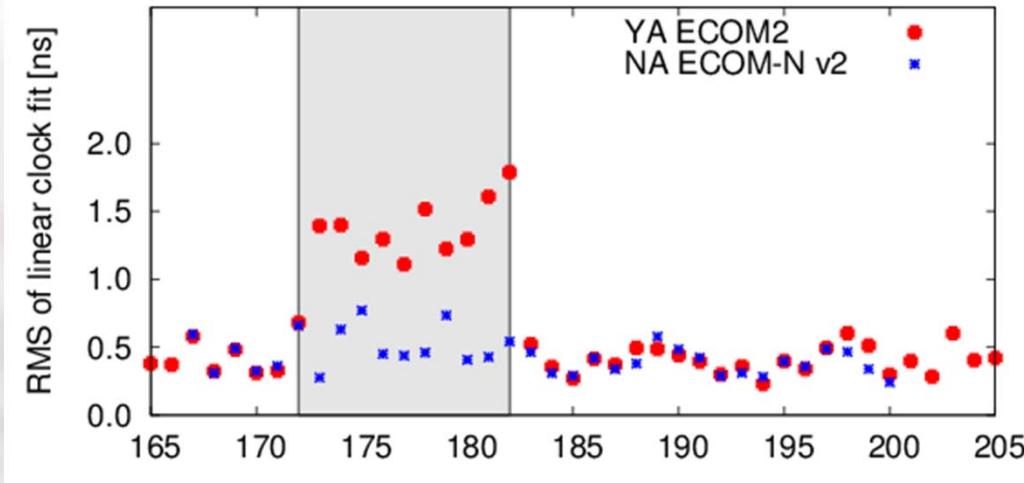
- ➡ Yaw-steering: no significant difference between ECOM versions
- ➡ Orbit normal attitude mode ($|\beta| < 4^\circ$, marked grey; wrong attitude considered): ECOM2 may even degrade solution

BeiDou clock validation



⇒ Experiments with ECOM versions better suited for orbit normal attitude mode

BeiDou clock validation



⇒ Experiments with ECOM versions better suited for orbit normal attitude mode



Summary

- Still a long way to go until new GNSS can contribute to CODE's IGS solutions with the same quality as GPS
- Our current focus: correct consideration of orbit normal attitude (challenges are, e.g., SRP modelling, detection and consideration of mode-transitions, ...)
- Next: use of published or own estimated transmitter antenna phase center corrections for Galileo, BeiDou, and QZSS
- Reprocessing of data from 2015 planned

Thank you
for
your interest!



Astronomical Institute University of Bern

