An Empirical Model of Electron Flux from the Seven-Year Van Allen Probe Mission

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May 4, 2020

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Background and Motivation

Desire data-based model on day-long timescales

- Empirical models have been designed to predict radiation environment (e.g., Roeder et al., Space Weather 2005; Chen et al., JGR, 2014), but they **may not capture actual fluxes observed a particular day**
- AE9 provides probability of occurrence (percentile levels) for flux and fluence averaged over different exposure periods—not meant to capture daily variations
- Effects that require shorter-term integrals of the outer radiation belt may need special attention when it comes to environmental assessments.
 - Spacecraft charging (DeForest, 1972; Olsen, 1983; Koons et al., 2006; Fennell et al., 2008)
- Practical example: GPS solar array current and voltage degrades faster than predicted by any model (e.g., Messenger et al., 2011)—Are we correctly modeling the radiation environment?

Figure: Black line is remaining factor of solar array current.

Red, green and blue lines: modeled or expected solar panel current including all known factors.

What is missing?



Background and Motivation

Use Van Allen Probe data to provide actual daily fluence estimates, mapped to secondary satellite



Image Credit: NASA

Model Background

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Obtaining equatorial pitch angle distributions

We want to know what electron fluxes are everywhere along the field line.

But we only know the fluxes for electrons that bounce past Van Allen Probes.



Model Background

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Model Background

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Modeling Pitch Angle Distributions

Legendre Polynomial Fitting vs. Filling the Gap with a Sine Function



Legendre Polynomial fitting may be good for statistics (e.g. Chen et al., 2014), but we want actual daily fluences Not for re-use. All Rights Reserved.

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Radiation Belt Daily Electron Flux Model: RB-Daily-E

Matrix: 0.2 R_E *L* shell (from 2-7 R_E) x 17 pitch angle

- 1440 minutes per day x 17 pitch angles x 25 energy bins are saved.
- Bin fluxes into 0.2 R_E L shell bins by summing total fluxes and normalizing by time spent in each bin that day.
 - Allows us to change L shell bin width without reprocessing.

Right: Example of 54, 226, and 1574 keV fluxes at 90, 121, 153, and 174 deg equatorial pitch angles. Must assume MLT symmetry.



GPS Fluxes from RB-Daily-E

Fly hypothetical GPS through model each minute by:

- 1) Mapping GPS location to equator (IGRF) to get L bin.
- 2) Integrate fluxes within pitch angle range of particles that reach GPS to get omnidirectional flux. (e.g. Local 0-180 at GPS may map to equatorial pitch angles of 0-60 and 120-180: see insert.)
- 3) Integrate omnidirectional flux over time to obtain a daily fluence as input to degradation models.



GPS Fluxes from RB-Daily-E

Integrated Fluxes over Mission Lifetime

- Plot fluences (time integrated fluxes)
 - These are inputs to solar cell degradation models
- Compare to AE9
 - AE9 Mean fluences (middle)
 - AE9 95th percentile (bottom)
- <1 means AE9 underestimates RBSP Model



GPS Fluxes from RB-Daily-E



Compare RB-Daily-E with Arase XEP Data



Summary

- We built a daily average electron flux model based on 7 years of Van Allen Probe (RBSP) data.
 - L shell x Pitch angle x Energy
 - 2 RE to 7 RE, > 7RE supplement with THEMIS statistics
 - 33 keV to 7.7 MeV
- RB-Daily-E provides daily average fluences for a given satellite providing ephemeris and date range as input.
- RB-Daily-E accurately predicted Arase fluences within a factor of ~1.2 for energies 600-1987 keV, and within a factor of ~2 for higher energies.
- RB-Daily-E performs more precisely than AE9 Mean or 95th percentile.
- Practical application: RB-Daily-E outputs can be used in solar cell degradation models.
 - We have had exciting results in this arena that we hope to share in the future.



Backup

Methodology Overview

Pitch Angle Distributions at the Equator

- Obtain equatorial pitch angle distributions (PADs) from Van Allen Probe data
 - Map Van Allen Probe satellites to equator
 - Determine equatorial PAD from local PAD
- Create daily flux averages for each L shell bin (2-7 RE with 0.2 RE increments), energy bin, and pitch angle bin.
- Fly GPS through the model to obtain daily fluences
 - Use GPS ephemeris to map GPS to equator
 - Determine which portion of the equatorial PAD will reach GPS
 - Integrate to obtain omnidirectional flux at GPS
 - Collect flux estimate every minute, and integrate over the day
- Use daily integrated fluxes as input to degradation models





Methodology

Mapping various pitch angles

- Because GPS could be at the equator when Van Allen Probes is far from the equator, we need to fill in the populations that Van Allen Probes did not see.
- Step 1: Convert local pitch angles to equatorial pitch angles.
 - Use the L shell and Beq data provided with the MagEIS product. Use REPT
 - Use MagEIS data without background correction.



$$\alpha_{eq} = \sin^{-1} \sqrt{\frac{B_{eq}}{B_{sc}}} \sin^2(\alpha_{sc})$$

InvBratio= $\frac{B_{eq}}{B_{sc}}$

There are some points when Beq > Bsc, so InvBratio >1.

This doesn't really mean anything, it must result from the data not following the model. (For example, dipolarization could mean observed B is larger than modeled B at the equator. Or a compression of the Magnetosphere could increase observed B.)

Because asin(1.5)=NaN, the mapped pitch angles are stored as NaNs when Beq>Bsc. They are interpolated over.

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Treated \alpha_{sc} as \alpha_{eq}.
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Methodology

Pitch angle model

To fill the gap, I use the four data points surrounding the gap and treat them as part of a sine function of the following form:

$$flux = A * \sin(B \alpha + \frac{\pi}{2} - B\frac{\pi}{2})$$

MPFIT uses the Levenberg-Marquardt technique to solve the least-squares problem.

Method to fill the gap:

- 1. Normalize the PAD by the maximum flux at that time and energy channel (so flux ranges from 0 to 1).
- 2. Determine if there is a decrease or an increase at 90 based on the four surrounding data points.
 - If there is a decrease,
 - Subtract the PAD by 1 (see example). Always want 0 and 180 to be near 0.
 - Set initial A parameter to -0.5. (Helps the fitting routine know where to start.)
 - If there is an increase,
 - Set initial A parameter to 1. (Helps the fitting routine know where to start.)



Black: "the data" to fit Navy: Normalized data 4 dots: the 4 normalized data points to fit Cyan: Fitted sine function.

Remember: Only need to fit the gap, the wings do not matter.

- 3. Run fitting routine, which fits sine function to 4 data points near gap.
 - Outputs: A and B.
- 4. Use A and B in Flux eqn to calculate flux for 80, 90, and 100 deg equatorial pitch angle.
- 5. Multiply by the maximum flux to "unnormalize" the mapped data.
- 6. Use real mapped data & values calculated in step 4. Interpolate to the 17 pitch angle values in REPT to get consistent equatorial pitch angle outputs.

Methodology Pitch angle model 3.0keV 03:59 2.0×10⁵ 1.5×10⁵ Real examples. Arrows point to 1.0×10⁵ the four data 5.0×10⁴ points used to C do the sine fit. 50 100 150 200 0 33.0keV 05:59:28 4×10⁵ 3×10⁵ 2×10⁵ 1×10⁵ Q 50 100 150 200 0 $flux = A * \sin(B \alpha + \frac{\pi}{2} - B \frac{\pi}{2})$



Applied Product (Results)

GPS daily flux counts over time, separated by energy

Knowing the equatorial fluxes, we fly a virtual GPS through the flux map to determine what fluences GPS observed.

- 1. Map GPS to equator to obtain its L shell and Beq. Initially use IGRF. (Now using T89)
- 2. Using Beq and Bsc, calculate maximum equatorial pitch angle that GPS observed.

$$\alpha_{eq1} = \sin^{-1} \sqrt{\frac{B_{eq}}{B_{sc}}} \sin^2(90^\circ) \qquad \alpha_{eq2} = 180 - \alpha_{eq1}$$
3. Integrate fluxes for all observed pitch angles:
$$j_{ommi}(E) = \frac{\int_{0}^{\pi} j(\alpha, E) \sin \alpha d\alpha}{\int_{0}^{\pi} \sin \alpha d\alpha}$$

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Where flux is a function of time (t), energy (W), and pitch angle (α).

Applied Product (Results)

GPS daily flux counts over time, separated by energy

Using the trapezoid rule, we can numerically integrate over the equatorial pitch angle distribution to get the omnidirectional flux at GPS.

$$j_{omni} = \frac{\sum_{n=0}^{m_1} (j_{n+1} + j_n) \left(-\cos(\alpha_{n+1}) + \cos(\alpha_n) \right) + \sum_{n=m_2}^{m_3} (j_{n+1} + j_n) \left(-\cos(\alpha_{n+1}) + \cos(\alpha_n) \right)}{(\cos(0) - \cos(\alpha_{m1})) + (\cos(\alpha_{m2}) - \cos(\pi))} \left(\frac{1}{2} \right)$$

Where m1 corresponds to the pitch angle bin α_{eq1} , the largest pitch angle below 90 that reaches GPS, m2 corresponds to the pitch angle bin α_{eq2} , the lowest pitch angle bin above 90 that reaches GPS, and m3 corresponds to the 180 (or π) pitch angle bin. We divide by 2 for trapezoid rule.

We multiply the above by 4π to integrate over steradians, and by 60 because GPS has a one-min cadence. Multiplying by 60 integrates over time. Units are #/cm^2-keV. (Easily converted to /MeV.)



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Trapezoid rule: find the area in the rectangle formed by (X2,Y2) and (X3, Y2). Add this area to half of the area of the rectangle formed by (X2,Y2) and (X3, Y3).

 $\begin{array}{l} J2^{*}(\alpha 3-\alpha 2)+(J3-J2)\ ^{*}(\alpha 3-\alpha 2)/2\\ ->(\alpha 3-\alpha 2)^{*}(J2+(J3-J2)/2)\\ ->(\alpha 3-\alpha 2)(1/2)\ ^{*}(2J2+J3-J2)\\ ->(\alpha 3-\alpha 2)(1/2)\ ^{*}(J2+J3)\\ & \text{Not for re-use. All Rights Reserved.} \end{array}$

Figure from "Dummies.com"

Runov et al. 2015 Results

Four R-bins from THEMIS statistics in the plasma sheet extends our flux model out to 25 R_F .





Figure 1. XY_{GSM} distribution of DFB/bubble events detected during the 2008 and 2009 THEMIS tail seasons. Colors indicate geocentric distance (magenta: $R < 9.5 R_E$, black: $9.5 \ge R < 12 R_E$, blue: $12 \ge R < 15.5 R_E$, and green: $15.5 \ge R < 25 R_E$).

Runov et al., AGU JGR, 2015

References

DeForest, S. E., "Spacecraft charging at synchronous orbit," J. Geophys. Res. 77: 3587–3611 (1972).

Fennell, J. F., J. L. Roeder, G. A. Berg, and R. K. Elsen (2008), HEO Satellite Framd and Differential Charging and SCATHA Low-Level Frame Charging

Gabrielse, C., Pinto, V., Nishimura, Y., Lyons, L., Gallardo-Lacourt, B., & Deng, Y. (2019). Storm time mesoscale plasma flows in the nightside high-latitude ionosphere: A statistical survey of characteristics. Geophysical Research Letters, 46, 4079–4088. <u>https://doi.org/10.1029/2018GL081539</u>

Koons, H., J. mazur, A. Lopatin, D. Pitchford, A. Bogorad, R. Herschitz (2006), Spatial and Temporal Correlation of Spacecraft Surface Charging in Geosynchronous Orbit, Journal of Spacecraft and Rockets, Vol. 43, No. 1, pp 178-185

Messenger, S., E. Jackson, J. Warner, R. Walters, T. Cayton, Y. Chen, R. Friedel, R. M. Kippen, and B. Reed, Correlation of Telemetered Solar Array Data With Particle Detector Data On GPS Spacecraft (2011), IEEE TRANSACTIONS ON NUCLEAR SCIENCE, VOL. 58, NO. 6, PP. 3118-3125

Higashio, N., T. Takashima, I. Shinohara, and H. Matsumoto, The extremely high-energy electron experiment (XEP) onboard the Arase (ERG) satellite, Earth, Planets and Space, 10.1186/s40623-018-0901-x, 2018

Olsen, R.C.; Purvis, C.K. Observations of charging dynamics. J. Geophys. Res. 1983, 88 (A7), 5657–5667

Runov, A., V. Angelopoulos, C. Gabrielse, J. Liu, D. L. Turner, and X.-Z. Zhou (2015), Average thermodynamic and spectral properties of plasma in and around dipolarizing flux bundles, J. Geophys. Res. Space Physics, 120, 4369–4383, doi:10.1002/2015JA021166.

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