

Cleaning magnetometer data using multi sensor configuration

Application to GEO-KOMPSAT-2A

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1 Introduction

- Variance analysis
- GEO-KOMPSAT-2A

2 SOSMAG concept

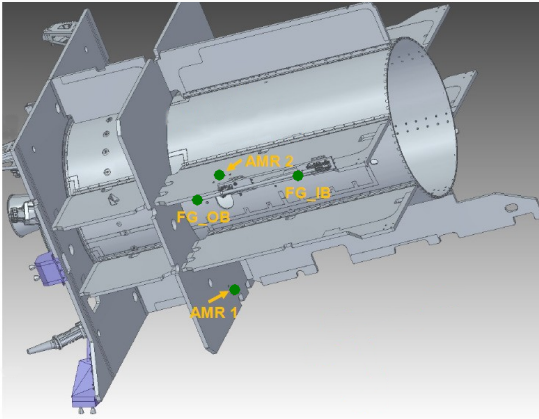
- Measurement configuration
- Cleaning algorithm

3 GK-2A magnetic field data cleaning

- Selected data
- Correction of FGMI and FGMO using AMR1
- Correction of FGM I/O using FGM O/I
- Results

It is a simple technique which identifies a more meaningful basis to represent the measured data

- assume N measured quantities, $q_i(t)$; $i = 1, \dots, N$ (e.g. $B_x(t), B_y(t), B_z(t)$)
 - Covariance matrix elements: $C_{ij} = \langle q_i q_j \rangle - \langle q_i \rangle \langle q_j \rangle$
 - diagonal elements represent the standard deviation σ^2 for each components
 - off-diagonal elements represent the correlation between different components
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- compute the eigenvalues / eigenvectors: $C \mathbf{v}_j = \lambda_j \mathbf{v}_j$; $j = 1, \dots, N$; $\lambda_1 > \lambda_2 > \dots > \lambda_N$
 - construct the transformation matrix to the new basis: $\mathcal{R} = (\mathbf{v}_1, \dots, \mathbf{v}_N)$
 - the transformed covariance matrix $C' = \mathcal{R}^T C \mathcal{R}$ is diagonal
 - most of the variation is in the first new component
 - correlations between the new components are minimized

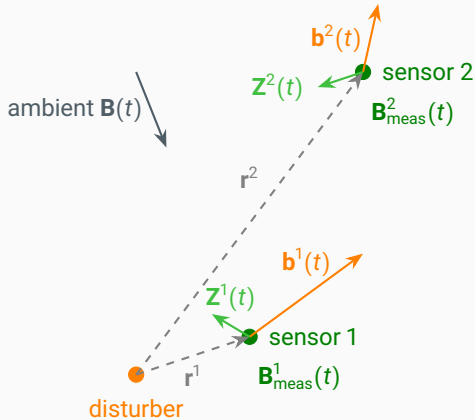


- Korean meteo / space weather satellite
- launched on Dec 4 2018
- geostationary orbit 128.2° E
- no magnetic cleanliness
- 2 FGM sensors on a 1 m long boom
- 2 AMR sensors on the spacecraft body

task:

- to remove the sc generated AC disturbances

Service Oriented Spacecraft Magnetometer (SOSMAG) concept



- Magnetic field measured at sensor i :

$$\mathbf{B}_{\text{meas}}^i(t) = \mathbf{B}(t) + \mathbf{b}^i(t) + \mathbf{Z}^i(t)$$

- $\mathbf{B}(t)$ → ambient field
- $\mathbf{b}^i(t)$ → disturbance at \mathbf{r}^i
- $\mathbf{Z}^i(t)$ → sensor specific noise and offset

- Difference between measurements:

$$\Delta \mathbf{B}_{\text{meas}}^{ij}(t) = \Delta \mathbf{b}^{ij}(t) + \Delta \mathbf{Z}^{ij}(t)$$

- Corrected measurement:

$$\mathbf{B}_{\text{corr}}^i = \mathbf{B}_{\text{meas}}^i + \mathcal{A}^{ij}(\Delta \mathbf{B}_{\text{meas}}^{ij}), \quad \mathcal{A}^{ij} = \alpha^{ij} \mathcal{R}(t)$$

Variance driven SOSMAG cleaning algorithm

Polarized disturbance: $\mathcal{R} = \text{constant in time} \Rightarrow$

- Transform to the **V**ariance **P**roduct **S**ystem (x-axis aligned with the maximum variance direction)
- Only the maximum variance component needs correction: $(B_{\text{corr}}^i)_x = (B_{\text{meas}}^i)_x - \alpha^{ij}(\Delta B_{\text{meas}}^{ij})_x$

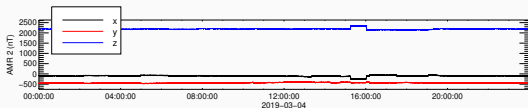
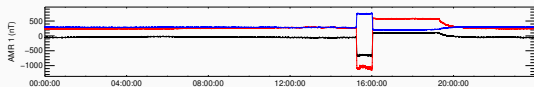
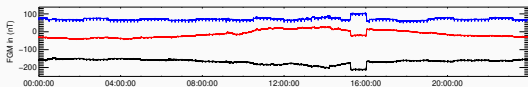
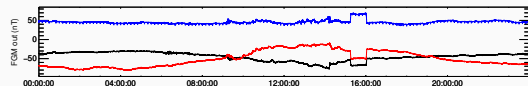
$\Rightarrow \mathcal{A}^{ij}$ is completely determined by the scalar coefficient α^{ij} and by the maximum variance directions:

$$\mathcal{A}_{kl}^{ij} = -\alpha^{ij} \left((\mathcal{R}^i)^{-1} \right)_{kx} \left(\mathcal{R}^{ij} \right)_{xl}, \quad \alpha^{ij} \text{ is estimated from the variance ratio}$$

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- Disturbance is removed but sensor specific is noise added: $(B_{\text{corr}}^i)_x = B_x + 0 + Z_x^i - \alpha^{ij}(Z_x^i - Z_x^j)$
 - Corrected measurements for the two sensors are identical: $(B_{\text{corr}}^i)_x \equiv (B_{\text{corr}}^j)_x$
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Higher order corrections:

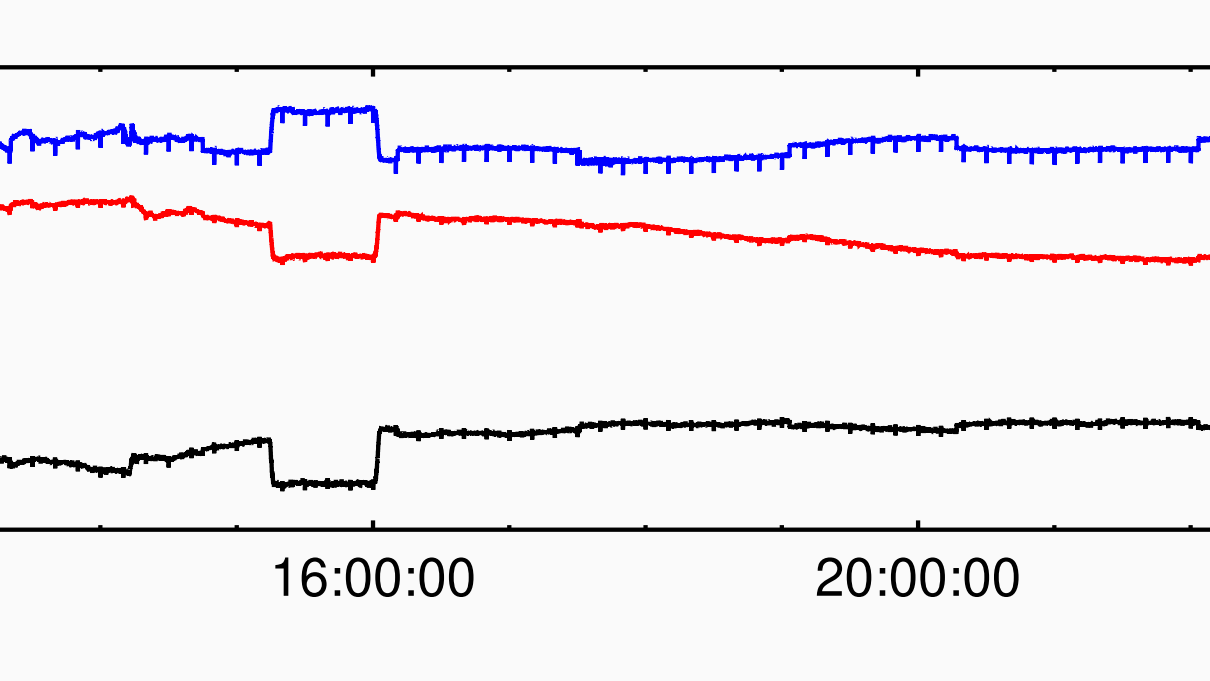
$$\mathbf{B}^{n,ij} = \mathbf{B}^{n-1,i} + \mathcal{A}^{n-1,ij} \Delta \mathbf{B}^{n-1,ij}, \quad \mathcal{A}_{kl}^{n-1,ij} = -\alpha^{n-1,ij} \left((\mathcal{R}^{n-1,i})^{-1} \right)_{kx} \left(\mathcal{R}^{n-1,ij} \right)_{xl}$$

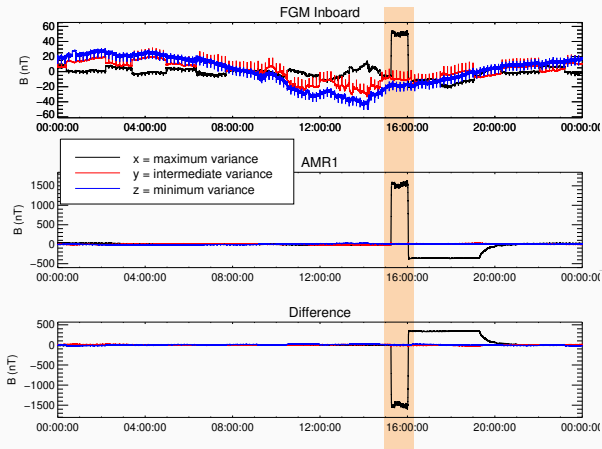


- Representative for the routine operations
- 16:00 disturbance seen by all sensors
- AMR/FGM amplitude ratios:
 - 40 for AMR 1
 - 5 for AMR 2

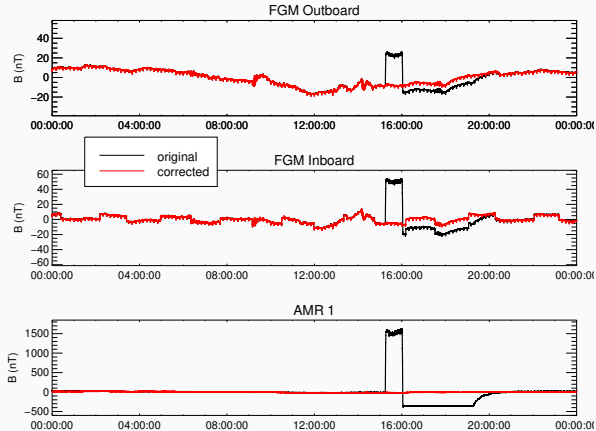
Strategy:

- Do not use the AMR2 data for AC cleaning
- 1st step: clean both FGMs using AMR1
- 2nd step: clean resulted FGMI/O using FGM0/I





- find the VPS for the [15:10,16:15] interval
- distinct coordinate systems
- rotate the entire day to the found VPS
- main disturbance only in x-component
- no steps in the z-component

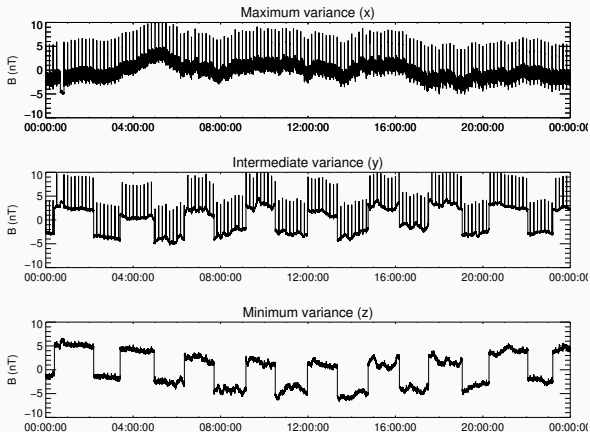


- scaling and rotation parameters:

- FGM Out $\alpha = 48$ $\theta = 51^\circ$ $\phi = 126^\circ$
- FGM In $\alpha = 27$ $\theta = 54^\circ$ $\phi = 141^\circ$
- AMR1 $\alpha = 0.97$ $\theta = 73^\circ$ $\phi = 114^\circ$

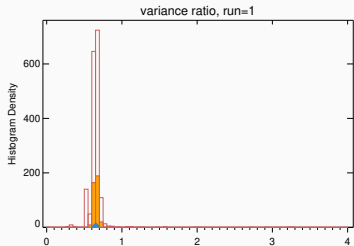
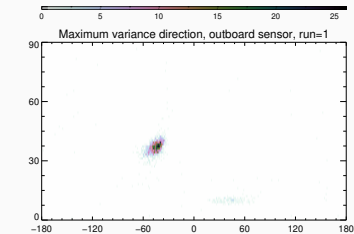
- targeted disturbance removed

- corrected FGM \Rightarrow 2nd step initial data

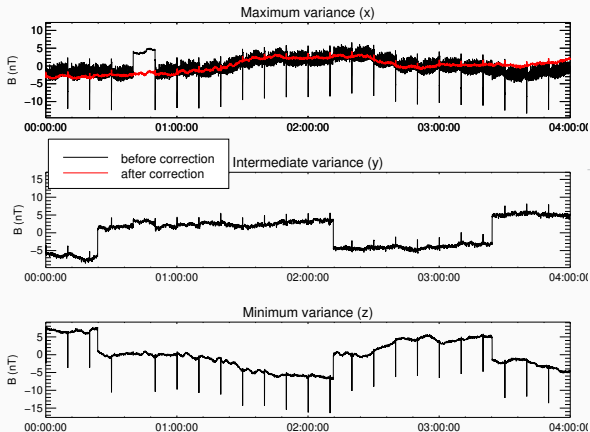


Disturbances decouple on components:

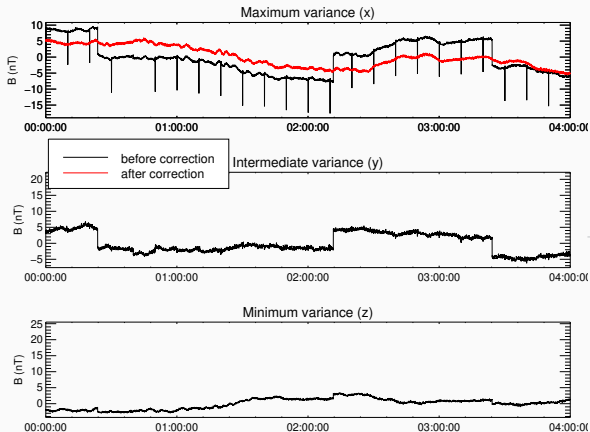
- high frequency noise on x
- spikes on x and y
- steps on y and z



- use sliding windows to determine the variance direction
 - determine the most probable variance direction
 - select the corresponding α scale factors
 - compute the mean scale factor
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- window size selects the disturbance to be removed
 - first order correction:
 - window size: 100 s (high frequency disturbance)
 - maximum variance direction: out $(-42^\circ, 37^\circ)$; in $(-38^\circ, 52^\circ)$
 - scale factor: out 0.653; in 1.620



- window size $w = 100 \text{ s}$
 - scale factor $\alpha = 1.6$
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- the HF noise is substantially reduced
 - the spikes are also reduced \Rightarrow
 - similar scale factors \rightarrow close sources
 - y-component noisier than z-component

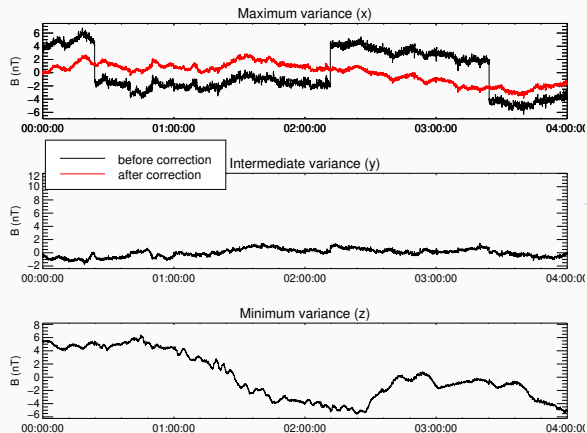


• window size $w = 700 \text{ s}$

• scale factor $\alpha = 1.4$

• the spikes are removed

• the steps are also removed



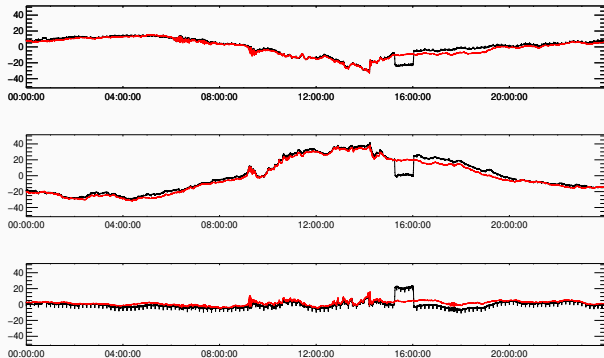
• window size $w = 16\,000\text{ s}$

• scale factor $\alpha = 1.1$

• the steps are removed

• remaining HF noise and spikes reduced

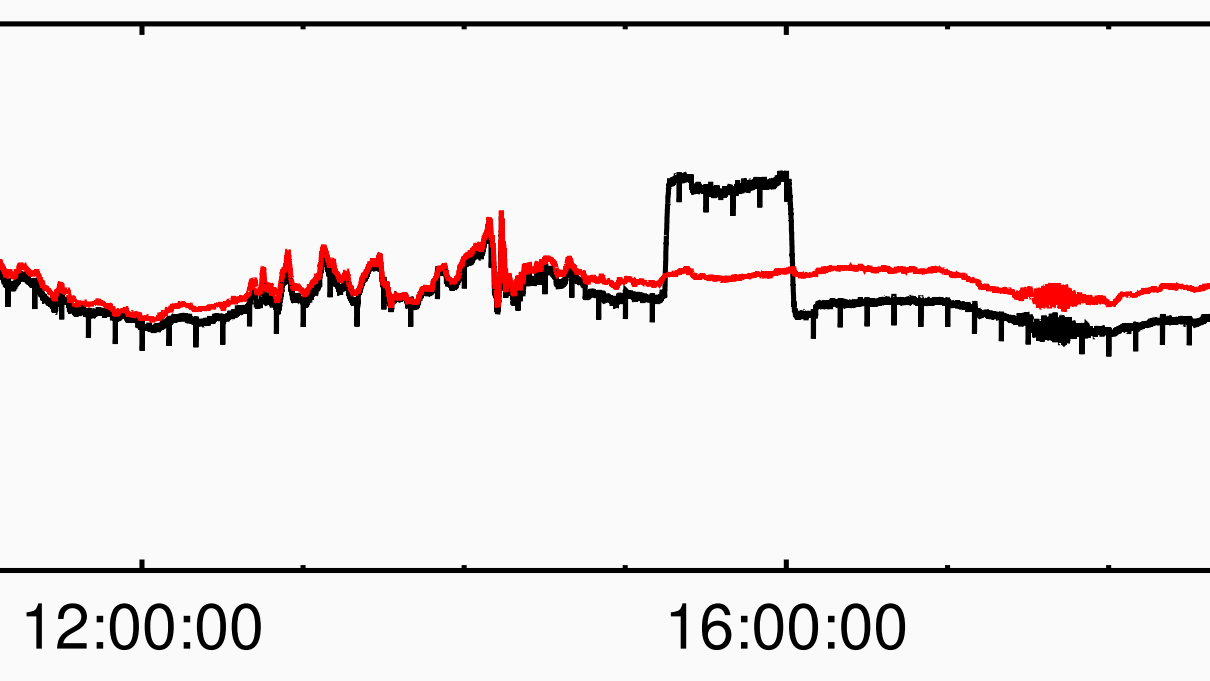
• remnants of steps and spikes on y

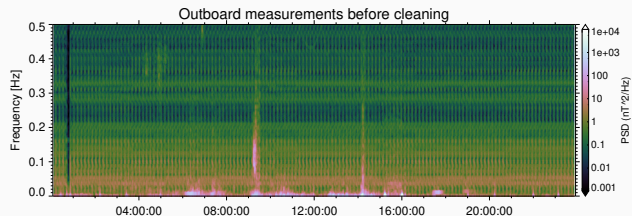


- combined single step correction:

$$\mathbf{B}_{\text{corr}}^0 = \mathcal{M}^0 \mathbf{B}_{\text{meas}}^0 + \mathcal{M}^I \mathbf{B}_{\text{meas}}^I + \mathcal{M}^{\text{AMR}} \mathbf{B}_{\text{meas}}^{\text{AMR}}$$

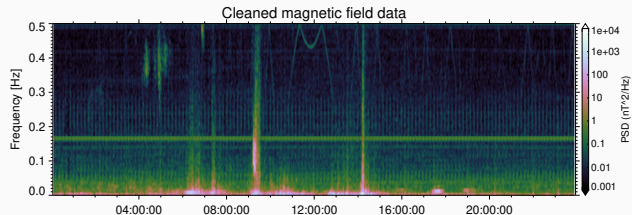
- parameters uploaded to the spacecraft
- cleaning is done now onboard





the spectral power of the disturbances is reduced in average by at least a factor of eight:

$$\text{mean}\left(\sum_{xyz} P_{\text{original}} / \sum_{xyz} P_{\text{cleaned}}\right) = 8$$



- variance analysis is a powerful tool for cleaning magnetometer data
 - multiple disturbances (time scales, directions) tend to decouple in the Variance Principal System
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- three from four available sensors on GK-2A were used to clean the data
 - the determined cleaning parameters were uploaded to the spacecraft