Langmuir circulation without wind or surface waves

Shear flow interacting with wavy topography

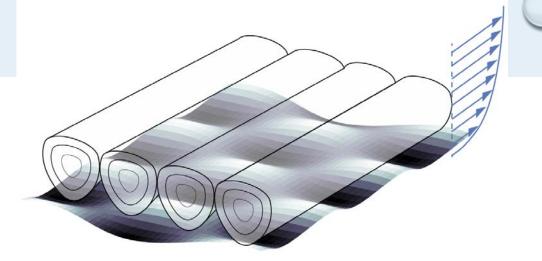
Andreas H. Akselsen, Andreas Brostrøm & Simen Å. Ellingsen

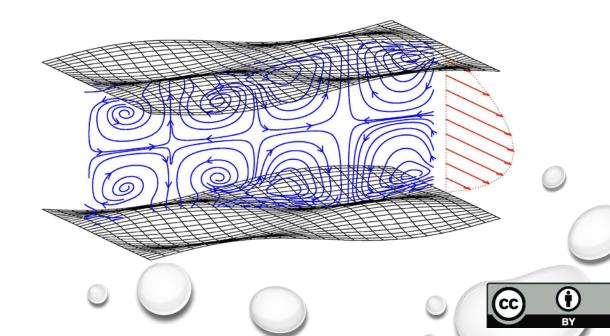




One-slide summary

- Streamwise vortices occur in fluid flow over surface with crossing-wave pattern.
- Kinematically equivalent to Langmuir circulation (the «CL1» mechanism, Craik 1970).
- Analytical theory presented for Langmuir rolls during
 - the early onset, and
 - the final, steady-state.
- Explicit expressions when velocity profile is a power law $U(z) = z^q$.
- Confirmed by Lattice-Boltzmann simulations.







Introduction

• Full details of this work may be found here: arXiv:2005.00317v1 [physics.flu-dyn]

Langmuir-type vortices in boundary layers driven by a criss-cross wavy wall topography

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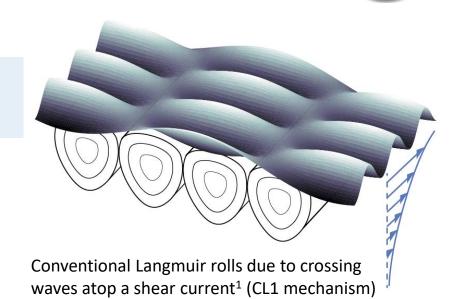
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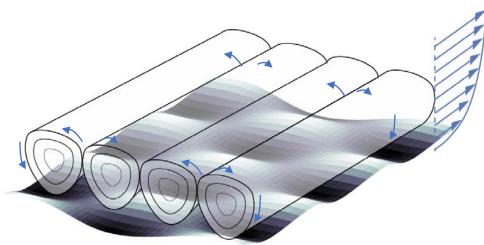




Introduction

- Conventional Langmuir circulation
 - Occurs due to interaction of waves and near-surface shear¹
 - Observable as «windrows» on surface, where foam etc gathers in downwelling areas.
 - Important contributor to mixing in the upper ocean².
- Neither wind nor surface waves are necessary
 - The Craik-Leibovich mechanisms are purely kinematic interaction between <u>mean shear</u> and <u>wavy fluid motion</u>.





Langmuir rolls due to boundary layer over a criss-cross wavy bottom. (Also CL1 mechanism)



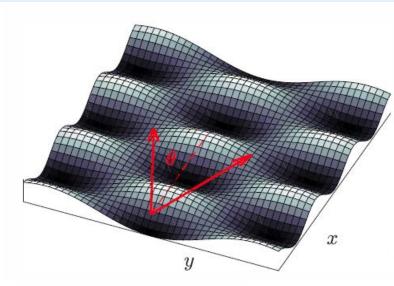
² S. E. Belcher *et al.*, *Geophys. Res. Lett.* **39** L18605 (2012)





Theory (outline)

- Follows roughly the procedure of Craik 1970¹. See our manuscript for full details.
- The wavy boundary introduces **perturbations** to the mean shear profile, treated up to **second order** in the wall corrugation steepness.
- Steps:
 - Assume known background flow U(z).
 - Derive (approximate) linear order perturbation based on a simplifying model
 - There are 4 second-order modes due to self-advection.
 - One of these modes is resonant, growing linearly with time. It has the form of longitudinal vortices, or «rolls».
- Some more details on the theory and derivation are found on later slides.



Boundary topography with wave-vectors $(k_x, \pm k_y)$ indicated.







Theory step 1: approximate linear solution

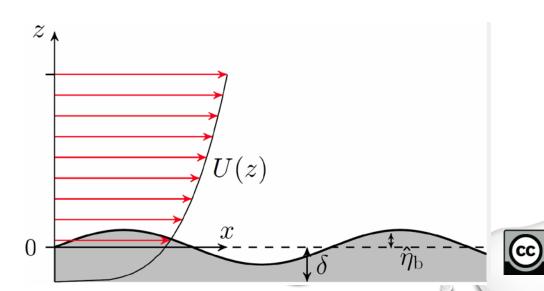
- Model: model the real (no-slip) wall by a displacement thickness of the same shape, creating a free slip, impermeable boundary.
 - Captures all essentials of this kinematic effect
- Treat first-order perturbation as steady and inviscid
 - Viscosity's primary effect is to create mean shear U(z) and displacement.
- Linearising Euler equation w.r.t. perturbations and eliminating velocities gives a Rayleigh equation for the first-order pressure perturbation $p_1(\boldsymbol{k},z) e^{i\boldsymbol{k}\cdot\boldsymbol{r}}$:

$$p_1'' - 2\frac{U'}{U}p_1' - k^2p_1 = 0$$

and boundary conditions at bottom/top of domain.

- Easily solved numerically. Analytical solution for power law case $\,U\,=\,z^q\,$
- 1st order velocities given by

$$u = -\frac{p_1}{U} - \frac{U'p_1'}{k_x^2U^2}, \quad v = -\frac{k_y p_1}{k_x U}, \quad w = \frac{ip_1'}{k_x U}.$$



Theory step 2: resonant 2nd order mode

analytically on previous slide.

Navier-Stokes equation at 2nd order reduces to

 $(\partial_t - Re^{-1}\nabla^2)\nabla^2 w = \mathcal{R}(z)~;$ Convective term, products (based on characteristic velocity and depth) 2nd order vertical velocity perturbation 5 of 1st order quantities. Found p_1 numerically or

$$\mathcal{R}(z) = 8 \frac{k_{1y}^2}{k_{1x}^2} \frac{U'}{U^3} \left[\left(k_{1x}^2 - k_{1y}^2 \right) p_1^2 + (p_1')^2 \right].$$

- Second order harmonics from sums & differences of wave vectors $(\pm k_x, \pm k_y)$.
- Modes with purely spanwise wave number
 are resonant. Other 2nd order harmonics are negligible.

• Solutions can be found in two cases:

1. Early onset, transient growth

• Initially the 2nd order motion is <u>transient</u> and <u>inviscid</u>.

2. Ultimate steady state

• Eventually vortices are stabilised by viscosity, reaching a viscous and steady state.





Theory result: early onset transient growth

- Initial growth assumed essentially inviscid; set Re⁻¹=0.
- Results in

$$w = \sum_{\pm} d^{\pm} e^{\pm \kappa z} + w_{\times}(z);$$

with particular solution growing <u>linearly</u> in time:

$$w_{\times}(z) = \frac{t}{\kappa} \int_{0}^{z} d\xi \, \mathcal{R}(\xi) \, \sinh \kappa (z - \xi).$$

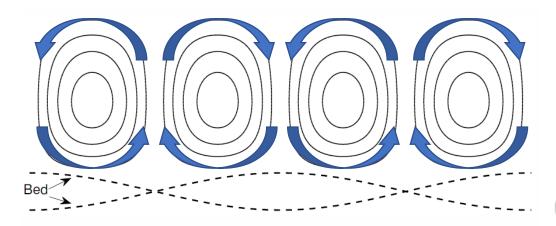
(d^{\pm} chosen to satisfy boundary conditions)

• Spanwise velocity:

$$v = i w' / \kappa_y$$

Streamwise velocity increases <u>quadratically</u> with t:

$$u(z,t) = -\frac{t}{2}U'(z)w(z,t),$$



Downwelling towards crest/trough-line, upwelling from saddlepoint-line







• Set transient term in Navier-Stokes to zero. Again a simple solution:

$$w = \sum_{\pm} (d_0^{\pm} + z d_1^{\pm}) e^{\pm \kappa z} + w_{\times}(z);$$

$$w_{\times}(z) = \frac{Re}{2\kappa^3} \int_0^z d\xi \, \mathcal{R}(\xi) \, G[\kappa(z - \xi)];$$
$$G(Z) = \sinh(Z) - Z \cosh(Z).$$

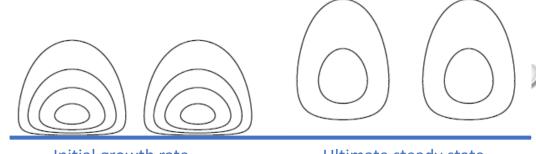
• As before, $v=\mathrm{i}\,w'/\kappa_y$, and we find the streamwise 2nd order velocity*

$$u(z) = \sum_{\pm} d_u^{\pm} e^{\pm \kappa z} + \frac{Re}{\kappa} \int_0^z d\xi \, U'(\xi) w(\xi) \sinh \kappa (z - \xi).$$

Note that spanwise and vertical velocities u, w scale as Re, but the streamwise velocity u scales as Re^2 .

• Initial growth closer to the wall; vortices moving towards the bulk before steady state.

Example: «deep» water (upper boundary far away):



Initial growth rate

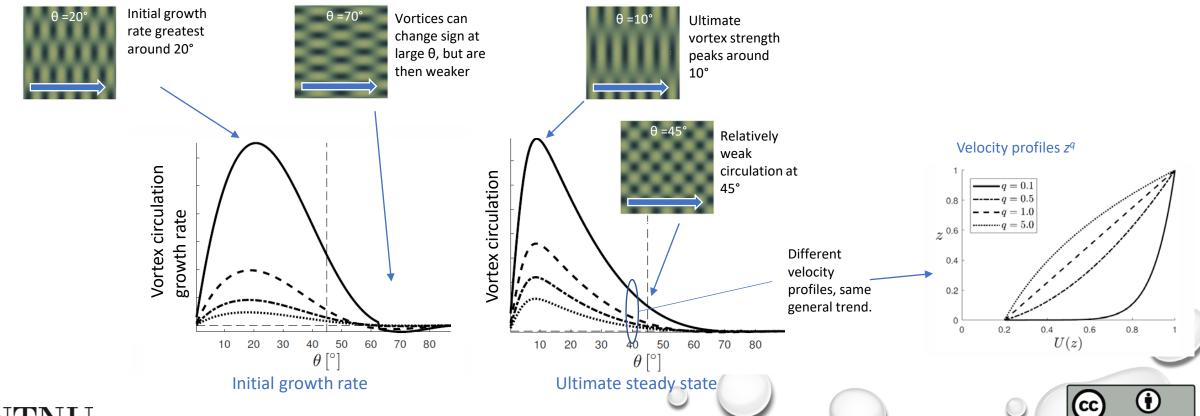
Ultimate steady state





Theory result: dependence on crossing angle θ

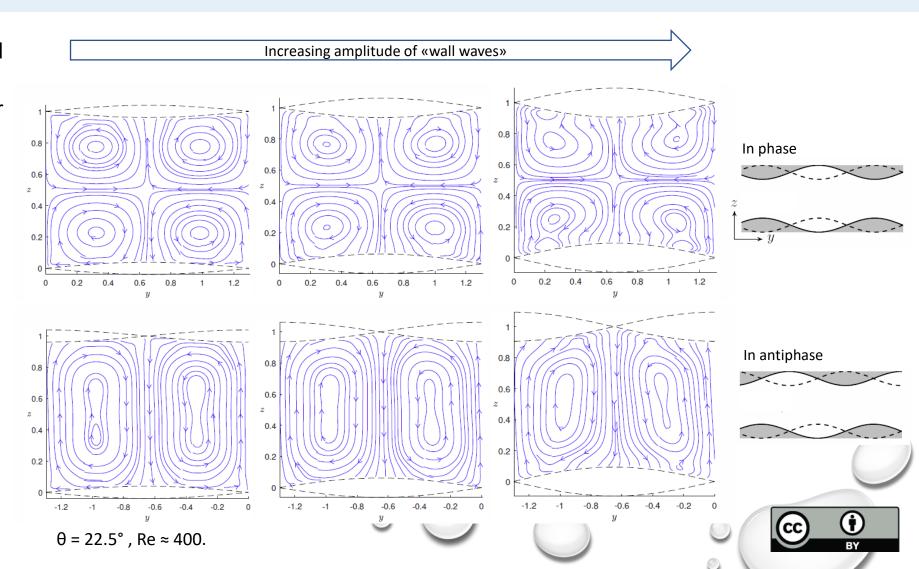
• Circulation strongest for «protracted eggcarton», $\theta \sim 10^{\circ}$ - 20°





Confirmation by numerical simulation

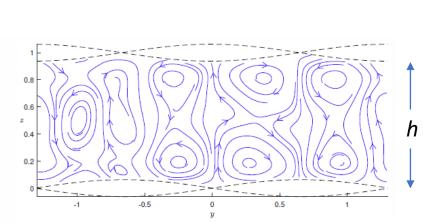
- The phenomenon is confirmed by simulation (Lattice-Boltzmann method) of laminar flow between two plates with crossing-wave pattern.
- Tested effect of amplitude, Re and spanwise phase shift.





Confirmation by numerical simulation

- The phenomenon is confirmed by **simulation** (Lattice-Boltzmann method) of laminar flow between two plates with crossing-wave pattern.
- Tested effect of amplitude, Re and spanwise phase shift.



Vortices become unstable at high Reynolds numbers. Here Re = 1600; Geometry: , $\vartheta = \pi/2$, $\theta = 22.5^{\circ}$, $h = \lambda/2$.

Flow snapshots: cross-flow plane Re = 225, ϑ = 0 Re = 900, ϑ = 0 Re = 225, $\vartheta = \pi/2$ Re = 900, $\vartheta = \pi/2$ Streamlines slices of full three-dimensional flow field. $a \approx 0.0625$, $h = \lambda/2$ $\theta \approx \pi/8$. Slice at $x = \pi/6k_x$.